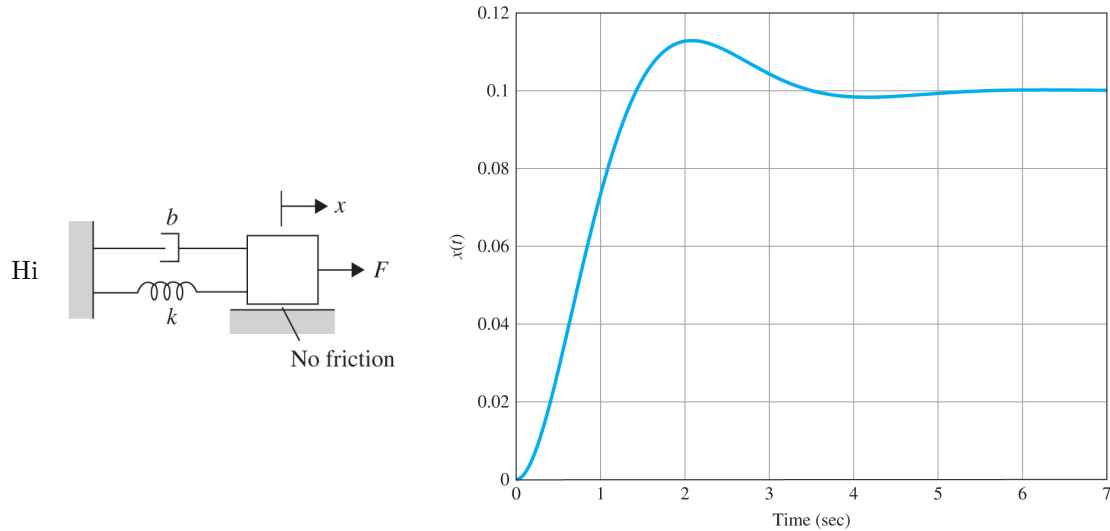


Control Systems : Set 4 : PID (3) - Solutions

Prob 1 | A simple mechanical system is shown in the figure below. The parameters are k = spring constant, b = viscous friction constant and m = mass. A step of 2 Newtons force is applied and the resulting step response is shown below. What are the values of the system parameters k , b , and m ?



The equation of motion for this system is

$$m\ddot{x} + b\dot{x} + kx = F$$

which gives the transfer function

$$\frac{X}{F} = G(s) = \frac{\frac{1}{m}}{s^2 + \frac{b}{m}s + \frac{k}{m}}$$

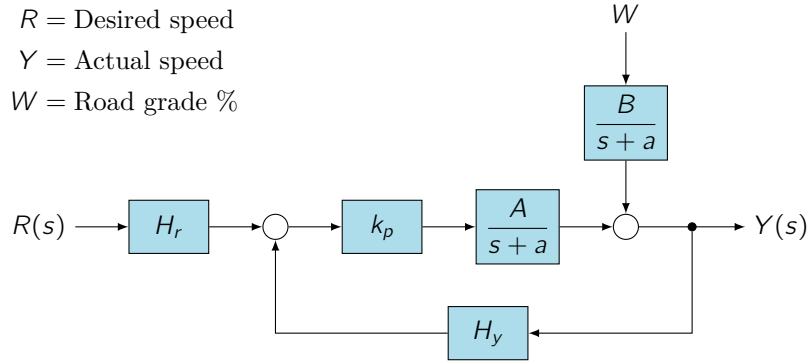
From the plot, we can estimate a number of properties:

DC gain	$2G(0) = 0.1$	$\rightarrow$	$G(0) = 0.05$
Overshoot	$\frac{x(T_p) - x(\infty)}{x(\infty)} = \frac{0.113 - 0.1}{0.1} = 13\%$	$\rightarrow$	$\zeta = 0.54$
Peak time	$T_p = 2$	$\rightarrow$	$\omega_n = 1.87$

We can now solve for the system parameters

$G(0) = 0.05 = \frac{1}{k}$	$\rightarrow$	$k = 20$
$\omega_n^2 = 3.4969 = \frac{k}{m}$	$\rightarrow$	$m = 5.72$
$2\zeta\omega_n = 2.0196 = \frac{b}{m}$	$\rightarrow$	$b = 11.55$

Prob 2 | Consider the automobile speed control system depicted in the figure below.



- a) Find the transfer functions from $W(s)$ and from $R(s)$ to $Y(s)$.

Start at Y and work backwards

$$\begin{aligned}
 Y &= \frac{B}{s+a}W + \frac{k_p A}{s+a}(H_r R - H_y Y) \\
 (s+a)Y &= BW + k_p A H_r R - k_p A H_y Y \\
 (s+a+k_p A H_y)Y &= BW + k_p A H_r R \\
 Y &= \frac{B}{s+(a+k_p A H_y)}W + \frac{k_p A H_r}{s+(a+k_p A H_y)}R
 \end{aligned}$$

So we see that the transfer functions are

$$\frac{Y}{W} = \frac{B}{s+(a+k_p A H_y)} \quad \frac{Y}{R} = \frac{k_p A H_r}{s+(a+k_p A H_y)}$$

- b) Assume that the desired speed is a constant reference r_o , so that $R(s) = r_o/s$. Assume that the road is level, so $w(t) = 0$. Compute values of the feedforward gain H_r to guarantee that

$$\lim_{t \rightarrow \infty} y(t) = r_o$$

Consider two cases

- (i) $H_y = 0$. This is an open-loop feed-forward controller, as there is no feedback.
 (ii) $H_y \neq 0$. This is now a closed-loop feedback controller.

The speed is

$$Y = \frac{k_p A H_r}{s+(a+k_p A H_y)}R = \frac{k_p A H_r}{s+(a+k_p A H_y)} \frac{r_o}{s}$$

We use the Final Value Theorem to compute the steady-state output

$$\begin{aligned} y_{ss} &= \lim_{t \rightarrow \infty} y(t) = \lim_{s \rightarrow 0} sY(s) = \lim_{s \rightarrow 0} s \frac{k_p A H_r}{s + (a + k_p A H_y)} \frac{r_o}{s} \\ &= \lim_{s \rightarrow 0} \frac{k_p A H_r}{s + (a + k_p A H_y)} r_o \\ &= \frac{k_p A H_r}{a + k_p A H_y} r_o \end{aligned}$$

(i) Case $H_y = 0$

$$y_{ss} = \frac{k_p A H_r}{a} r_o$$

Therefore in order for y_{ss} to equal r_o , we must choose $H_r = \frac{a}{A k_p}$.

(ii) Case $H_y \neq 0$

$$y_{ss} = \frac{k_p A H_r}{a + k_p A H_y} r_o$$

Therefore in order for y_{ss} to equal r_o , we must choose $H_r = \frac{a + k_p A H_y}{k_p A}$.

- c) Now assume that a constant grade disturbance $W(s) = w_o/s$ is present in addition to the reference input. Find the variation in speed Y due to the grade change for both the feed-forward and feedback cases, using the values for H_r computed in part (b). Use your results to explain (i) why feedback control is necessary and (ii) how the gain k_p should be chosen to reduce steady-state error.

From (a) we see that the output will be

$$Y = \frac{B}{s + (a + k_p A H_y)} \frac{w_o}{s} + \frac{k_p A H_r}{s + (a + k_p A H_y)} \frac{r_o}{s}$$

Applying the Final Value Theorem, we get

$$\begin{aligned} y_{ss} &= \lim_{s \rightarrow 0} \frac{B}{s + (a + k_p A H_y)} w_o + \frac{k_p A H_r}{s + (a + k_p A H_y)} r_o \\ &= \frac{B}{a + k_p A H_y} w_o + \frac{k_p A H_r}{a + k_p A H_y} r_o \end{aligned}$$

(i) Case $H_y = 0$ and $H_r = \frac{a}{A k_p}$.

$$\begin{aligned} y_{ss} &= \frac{B}{a + k_p A H_y} w_o + \frac{k_p A H_r}{a + k_p A H_y} r_o \\ &= \frac{B}{a} w_o + r_o \end{aligned}$$

We see that the variation in y_{ss} is $\frac{B}{a}$. This value is completely unaffected by the feedforward term, and therefore we cannot change the impact of the disturbance on the output.

(ii) Case $H_y \neq 0$ and $H_r = \frac{a+k_pAH_y}{k_pA}$.

$$\begin{aligned} y_{ss} &= \frac{B}{a+k_pAH_y}w_o + \frac{k_pAH_r}{a+k_pAH_y}r_o \\ &= \frac{B}{a+k_pAH_y}w_o + r_o \end{aligned}$$

We can now reduce the impact of the disturbance by choosing a larger value for the proportional control term k_p .

- d) Assume that $w(t) = 0$ and that the gain A undergoes the perturbation $A + \delta A$. Determine the error in speed due to the gain change for both the feed forward and feedback cases (use the values for H_r derived in (b)). How should the gains be chosen in this case to reduce the effects of δA ?

Note that the controller gains have to be chosen as a function of A , and not of $A + \delta A$, as the control engineer does not know that the system gains will change during operation.

(i) Case $H_y = 0$, $H_r = \frac{a}{Ak_p}$.

$$\begin{aligned} y_{ss} &= \frac{k_p(A + \delta A)H_r}{a}r_o \\ &= \frac{k_p(A + \delta A)\frac{a}{Ak_p}}{a}r_o \\ &= \frac{A + \delta A}{A}r_o \\ &= \left(1 + \frac{\delta A}{A}\right)r_o \end{aligned}$$

So we see that if there is a 5% change in A , we get a 5% change in output speed.

(ii) Case $H_y \neq 0$, $H_r = \frac{a+k_pAH_y}{k_pA}$.

$$\begin{aligned} y_{ss} &= \frac{k_p(A + \delta A)H_r}{a+k_p(A + \delta A)H_y}r_o \\ &= \frac{k_p(A + \delta A)\frac{a+k_pAH_y}{k_pA}}{a+k_p(A + \delta A)H_y}r_o \\ &= \frac{(A + \delta A)(a+k_pAH_y)}{A(a+k_p(A + \delta A)H_y)}r_o \\ &= \left(1 + \frac{a\delta A}{A(a+k_pA(A + \delta A)H_y)}\right)r_o \end{aligned}$$

So we see that the sensitivity of the steady-state velocity to changes A is

$$\frac{a}{a + k_p A(A + \delta A)H_y} \frac{\delta A}{A}$$

which can be made small by choosing the proportional gain k_p large.

Prob 3 | The open-loop transfer function of a unity feedback system (i.e., a system whose controller is a proportional gain equal to one) is

$$G(s) = \frac{K}{s(s+5)}$$

The desired system response to a step input is specified as having a peak time less than $t_p = 2\text{sec}$ and an overshoot less than $M_p = 10\%$.

- a) Determine whether both specifications can be met simultaneously by selecting the right value of K .

Closed-loop transfer function

$$T(s) = \frac{K}{s^2 + 5s + K} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

From which we see that

$$\begin{aligned} \omega_n^2 = K & \rightarrow \omega_n = \sqrt{K} \\ 5 = 2\zeta\omega_n & \rightarrow \zeta = \frac{5}{2\sqrt{K}} \end{aligned}$$

Overshoot of 10% gives:

$$\zeta = \frac{5}{2\sqrt{K}} \geq -\frac{\ln 0.1}{\sqrt{\ln^2 0.1 + \pi^2}} = 0.6 \rightarrow K \leq 17.36$$

Peak-time of 2 seconds gives:

$$\begin{aligned} T_p &= \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}} = \frac{\pi}{\sqrt{K} \sqrt{1 - \frac{25}{4K}}} = \frac{2\pi}{\sqrt{4K - 25}} \leq 2 \\ K &\geq \frac{\pi^2 + 25}{4} = 8.72 \end{aligned}$$

Therefore both conditions can be met if $8.72 \leq K \leq 17.36$.

- b) Sketch the associated region in the s -plane where both specifications are met, and indicate what root locations are possible for some likely values of K .

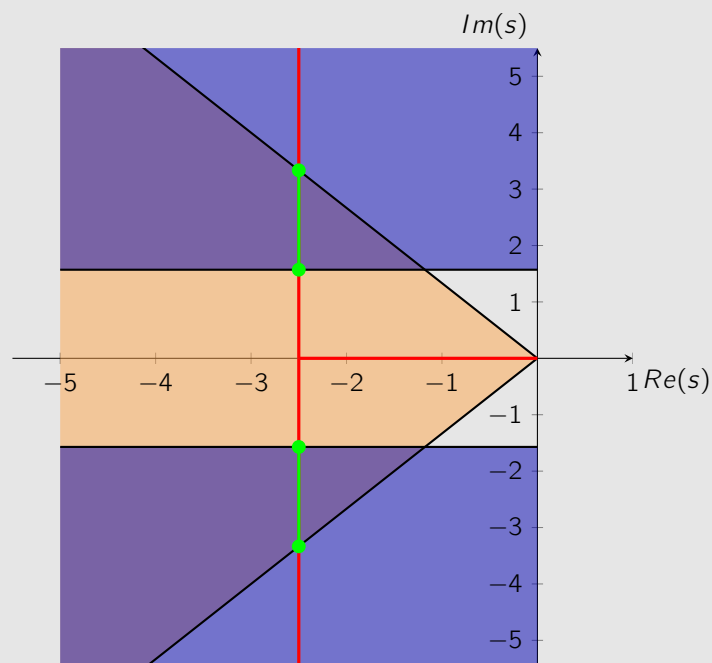
We can plot the constraints on the damping ratio and the natural frequency in the shaded region in the figure below.

The closed-loop poles can be calculated as a function of K by solving the characteristic equation

$$s^2 + 5s + K = 0 \quad \rightarrow \quad s = \frac{-5 \pm \sqrt{25 - 4K}}{2} = -2.5 \pm \sqrt{6.25 - 4K}$$

$$= \begin{cases} -2.5 \pm \sqrt{6.25 - K} & K \leq 6.25 \\ -2.5 \pm j\sqrt{K - 6.25} & K \geq 6.25 \end{cases}$$

The resulting curve is shown in red on the figure below, with the portion satisfying the constraints shown in green ($8.72 \leq K \leq 17.36$).



- c) Find the maximum value for K for the system to oscillate.

The system oscillates if the damping ratio is less than one.

$$\omega_n = \sqrt{K}$$

$$5 = 2\zeta\omega_n \quad \Rightarrow \quad 1 \geq \zeta = \frac{2.5}{\omega_n} = \frac{2.5}{\sqrt{K}} \quad \Rightarrow \quad K \geq 6.25$$